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PARAMETER IDENTIFICATION USING MODAL DATA

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PARAMETER IDENTIFICATION USING EXPERIMENTAL MODAL DATA

Introduction: Parameter Identification

- (1) Direct Approach - using actual sensor measurements to obtain parameters (e.g., ERA, ITD)
- (2) Indirect Approach - identifies parameters using measured modal data*

* in this work, only the natural frequencies are used to identify physical parameters

Equations of Motion

The motion of a distributed structure with zero damping is governed by a PDE of the form

$$\mathcal{L} u(P, t) + m(P) \ddot{u}(P, t) = f(P, t), \quad P \in D$$

$$B_i u(P, t) = 0, \quad P \in S \quad i = 1, 2, \dots, p$$

u = displacement at position P at time t

m = mass distribution

\mathcal{L} = stiffness differential operator of order $2p$

f = external force density

B_i = homogeneous differential operators of order ranging from zero to $2p-1$

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Discretization

Rayleigh-Ritz method: $u(P, t) = \underline{\psi}^T(P) \underline{q}(t)$

$\underline{\psi}(P)$ - n-vector of admissible functions

$\underline{q}(t)$ - n-vector of generalized coordinates

$$\ddot{M}\underline{q}(t) + K\underline{q}(t) = 0$$

$$M = \int_D m(P) \underline{\psi}(P) \underline{\psi}^T(P) d\Omega, \quad K = \int_D \underline{\psi}(P) \underline{\psi}^T(P) d\Omega$$

Discretization (cont'd)

Rayleigh-Ritz Type Parameter Expansion: $m(P) = \sum_{r=1}^g \alpha_r m_r(P)$

$$L = \sum_{r=1}^h \beta_r \lambda_r$$

$$M = \sum_{r=1}^g \alpha_r M_r, \quad K = \sum_{r=1}^h \beta_r K_r$$

$$M_r = \int_D \underline{\psi}(P) m_r(P) \underline{\psi}^T(P) dD = r\text{th mass matrix}$$

$$K_r = \int_D \underline{\psi}(P) L_r \underline{\psi}^T(P) dD = r\text{th stiffness matrix}$$

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Iterative Approach Using Frequency Method

$$q(t) = \sum_{r=1}^n x_r e^{i\omega_r t}$$

$$\omega_r^2 M_{0r} x_r = K_{0r} x_r \quad (\text{Actual Model})$$

$$\omega_{0r}^2 M_{0r} x_{0r} = K_{0r} x_{0r} \quad (\text{Postulated Model})$$

To refine the parameters a_{0r} , b_{0r} , we update them such that the theoretical natural frequencies converge to the measured natural frequencies of the actual structure.

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Postulated Parameters: $P_0 = [a_{01} \ a_{02} \ \dots \ a_{0g} \ b_{01} \ \dots \ b_{0h}]^T$

Measured Natural Frequencies: $\omega = [\omega_1 \ \omega_2 \ \dots \ \omega_f]^T$

Postulated Natural Frequencies: $\omega_0(P_0) = [\omega_{01} \ \omega_{02} \ \dots \ \omega_{0f}]^T$

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Iterative Approach (cont'd)

Sensitivity Analysis: $\Delta \tilde{\omega} = \frac{\partial \tilde{\omega}}{\partial \tilde{p}} \Delta \tilde{p}$, $\left[\frac{\partial \tilde{\omega}}{\partial \tilde{p}} \right] = \left[\frac{\partial \tilde{\omega}_i}{\partial p_i} \right]$

$$\Delta \tilde{\omega} = \tilde{\omega} - \tilde{\omega}(p_0), \quad \Delta \tilde{p} = \tilde{p} - p_0$$

\tilde{p} = updated parameters

Least-Squares Solution:

$$\Delta \tilde{p} = \left(\left[\frac{\partial \tilde{\omega}}{\partial \tilde{p}} \right]^T \left[\frac{\partial \tilde{\omega}}{\partial \tilde{p}} \right] \right)^{-1} \left[\frac{\partial \tilde{\omega}}{\partial \tilde{p}} \right]^T \Delta \tilde{\omega}$$

Iterative Approach (cont'd)

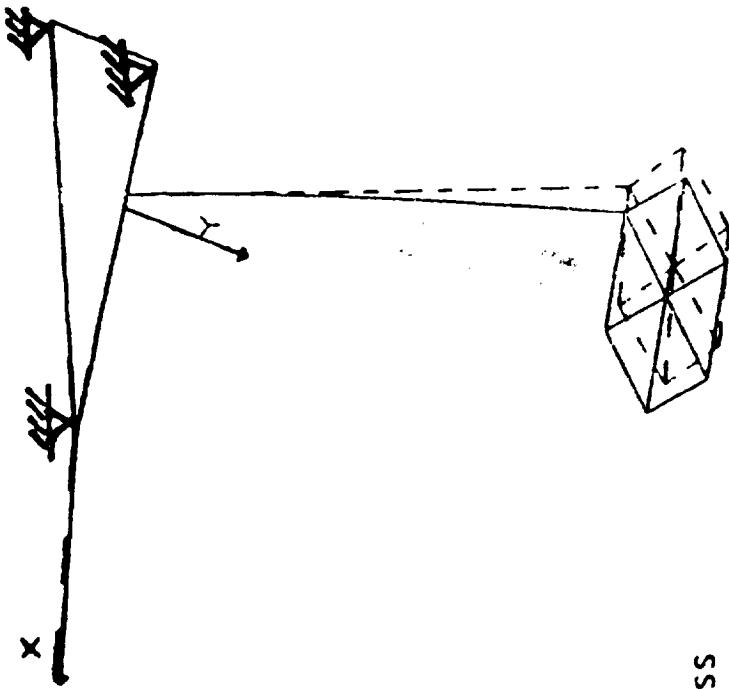
$$\text{Jacobian} \left(\frac{\partial \omega}{\partial p_i} \right) = \frac{\partial \omega_r}{\partial p_i} = \frac{1}{\kappa_r} \left[x_r^T \frac{\partial K}{\partial p_i} x_r \right]$$

$$\frac{\partial M}{\partial p_i} = \kappa_i, \quad \frac{\partial K}{\partial p_i} = 0 \quad (i = 1, 2, \dots, g)$$

$$\frac{\partial M}{\partial p_i} = 0, \quad \frac{\partial K}{\partial p_i} = \kappa_{i-g} \quad (i = g+1, g+2, \dots, g+h)$$

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Numerical Example

SCOLE MODEL:

Parameters:

m_A = antenna mass

I_{xA} = antenna moment of inertia with respect to roll axis

m = mast mass density

EI = mast bending rigidity

m_i = sensors and actuators at $z = -41.0, -87.8$ ($i = 1, 2$)

I_{xi} = additional mass moment of inertia ($i = 1, 2$)

Numerical Example

SCOLE M01EL

Kinetic Energy:

$$T(t) = \frac{1}{2} \int_0^L m_i \dot{w}_i^2(z, t) dz + \frac{1}{2} \sum_{i=1}^2 \{ [m_i \dot{w}_i^2(z_i, t) + I_{x_i} [\dot{w}_i'(z_i, t)]^2] \\ + \frac{1}{2} m_A \dot{w}^2(L, t) + \frac{1}{2} I_{x_A} [\dot{w}'(L, t)]^2 \}$$

Potential Energy:

$$V(t) = \frac{1}{2} \int_0^L \{ EI [w''(z, t)]^2 + P(z) [w'(z, t)]^2 \} dz$$

$w(z, t)$ - transverse displacement from equilibrium position

$P(z)$ = axial load

Numerical Example

- We used an $n = 4$ degree of freedom model:
- $\ddot{Mq} + \ddot{Kq} = \ddot{F}$
- Natural frequencies agree well with those obtained experimentally
(data provided by Lee, Williams and Sparks)
- We used the first 2 natural frequencies to update the bending rigidity EI and the mass moment of inertia of the antenna.

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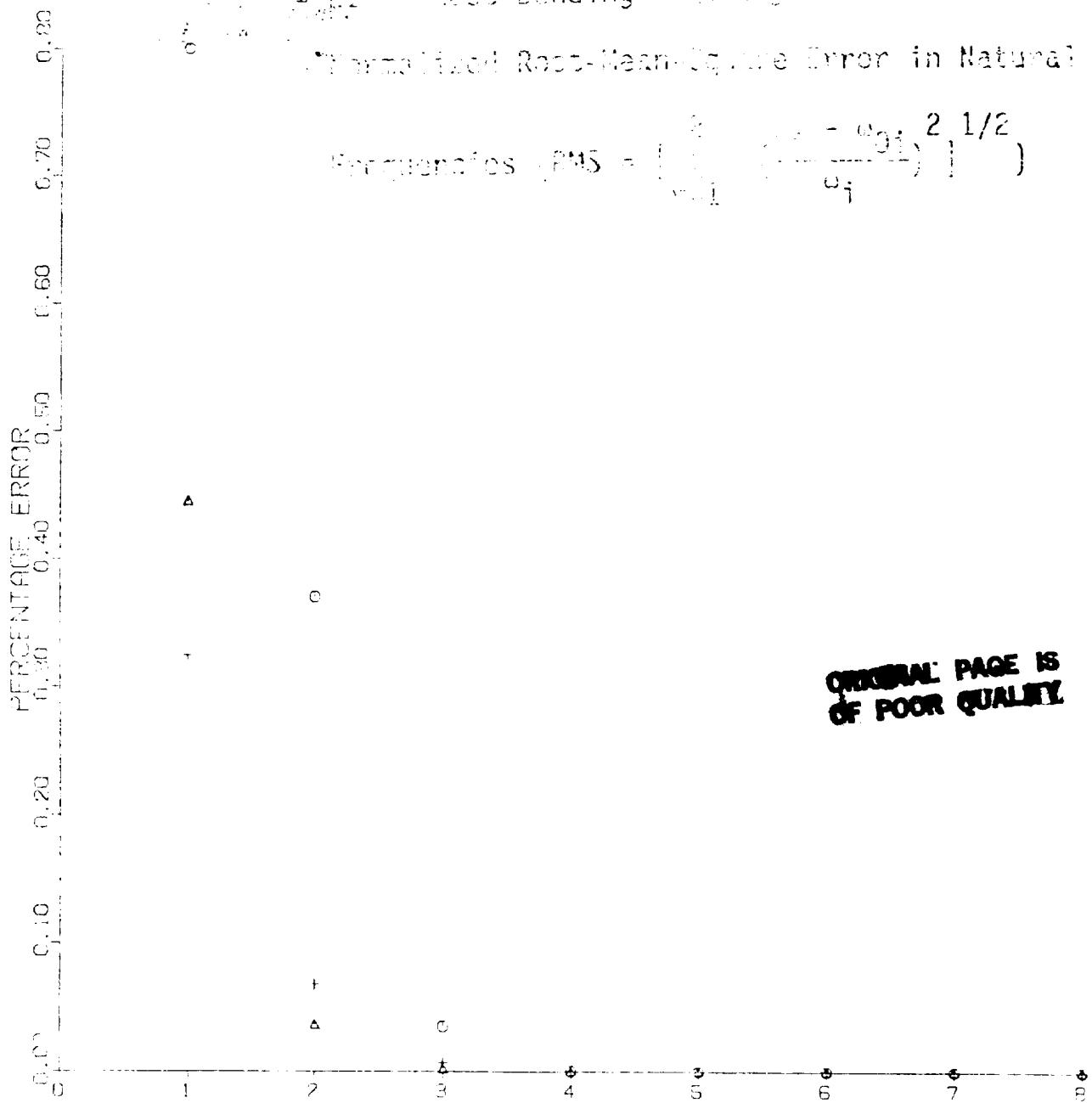
se I: Zero Error in Natural Frequencies

- I_{XA} Antenna Mass Moment of Inertia

- σ_{ω_i} Root Banding Mobility

- σ_{ω_i} Normalized Root-Mean-Square Error in Natural

$$\text{Frequencies RMS} = \left[\frac{1}{\omega_i} \left(\frac{\omega_i - \omega_{0i}}{\sigma_{\omega_i}} \right)^2 \right]^{1/2}$$



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Case III: 1% Error in ω_1

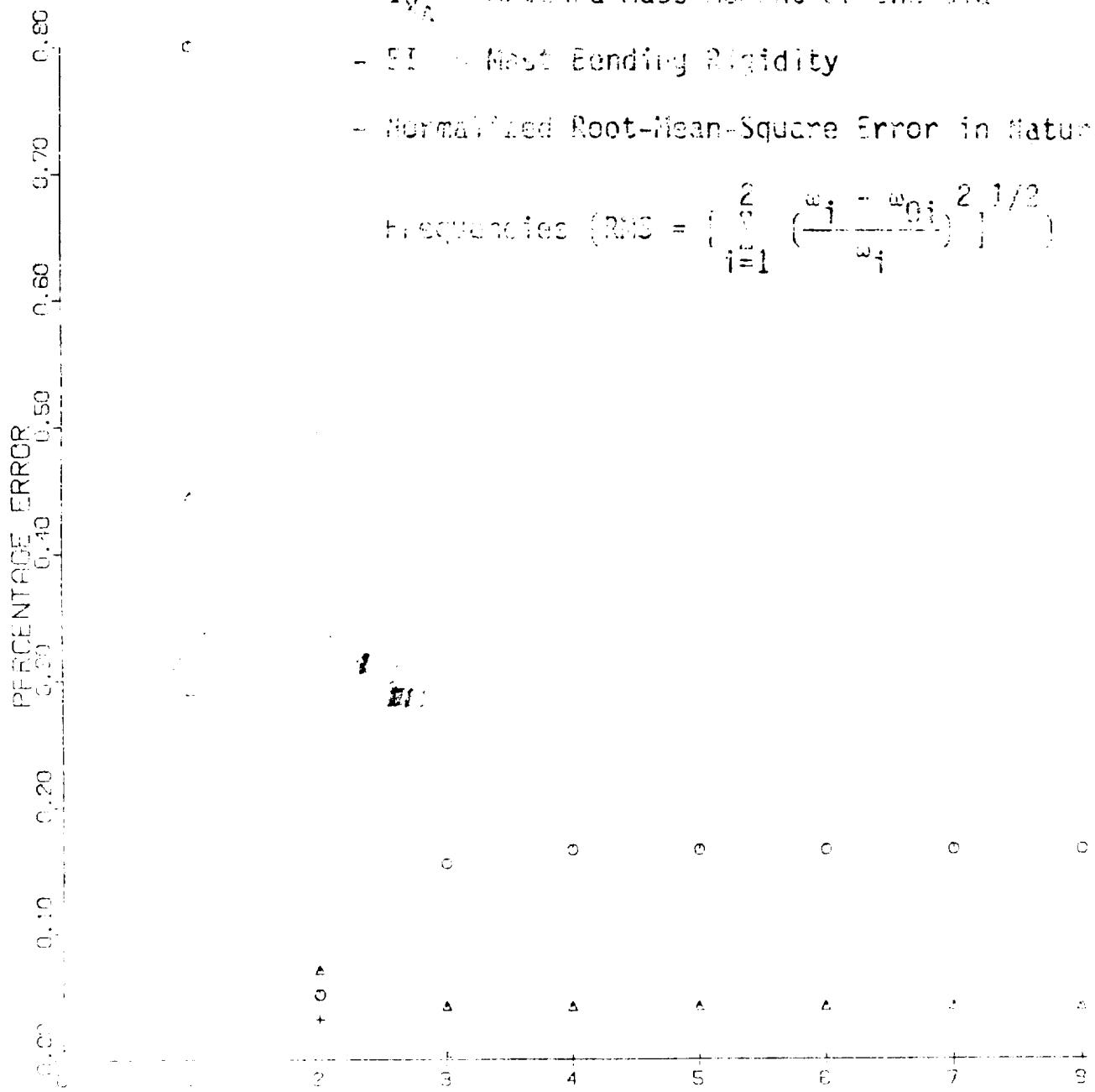
5% Error in ω_2

- I_{y_A} = Arienna Mass Moment of Inertia

- EI = Least Bending Rigidity

- Normalized Root-Mean-Square Error in Natural

$$\text{Frequencies (RMS)} = \left(\sum_{i=1}^2 \left(\frac{\omega_i - \omega_{0i}}{\omega_i} \right)^2 \right)^{1/2}$$



Case III: 2% Error in ω_1

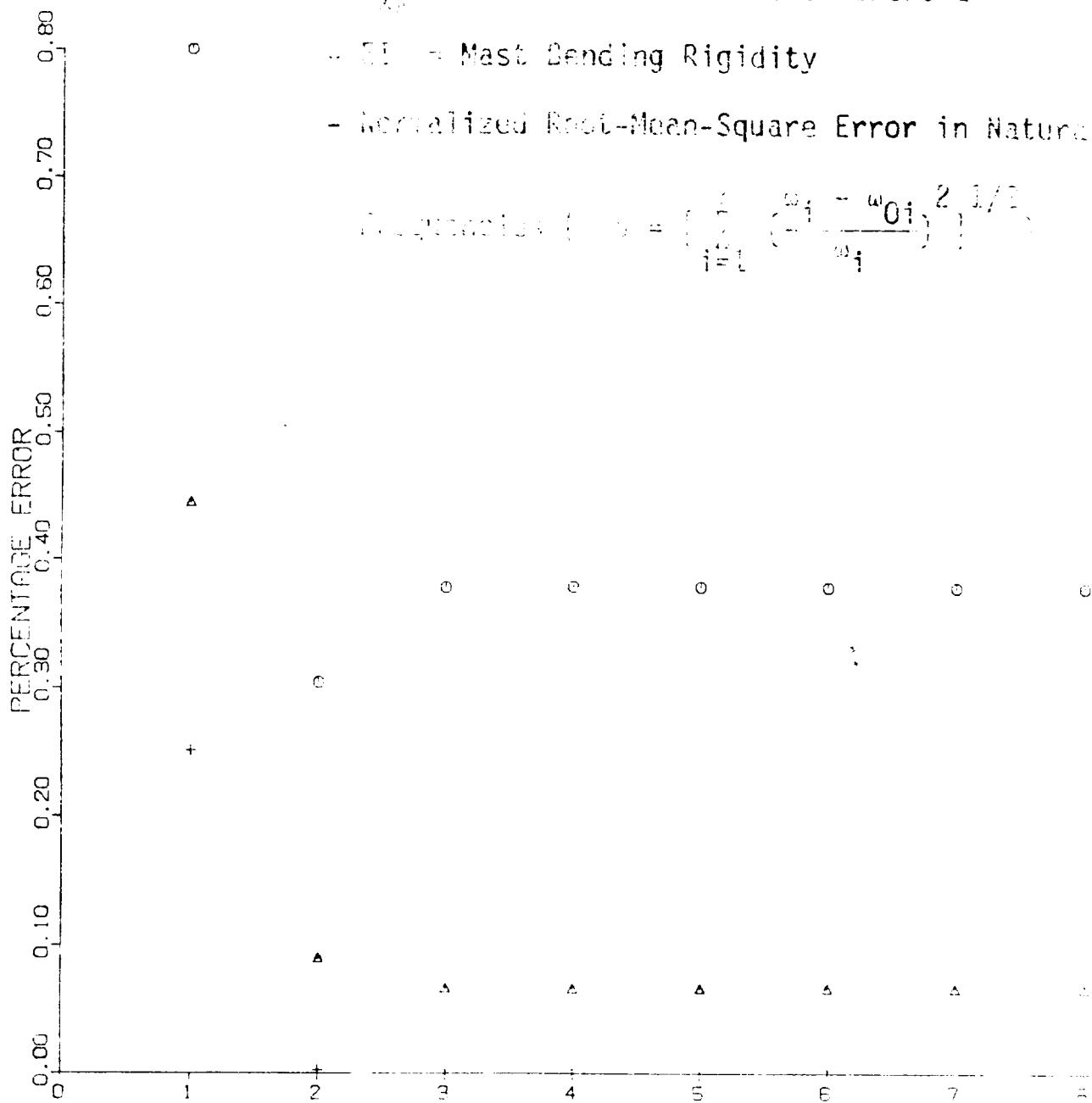
10% Error in ω_2

- I_{x_0} = Antenna Mass Moment of Inertia

○ - EI = Mast Bending Rigidity

- Normalized Root-Mean-Square Error in Natural

$$\text{Normalized Error} = \left(\frac{1}{n} \sum_{i=1}^n \left(\frac{\omega_i - \omega_{0i}}{\omega_i} \right)^2 \right)^{1/2}$$



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Conclusions

- Indirect approach is used to obtain physical parameters
- Results indicate that the parameters converge quickly
- The algorithm is relatively insensitive to noise

